

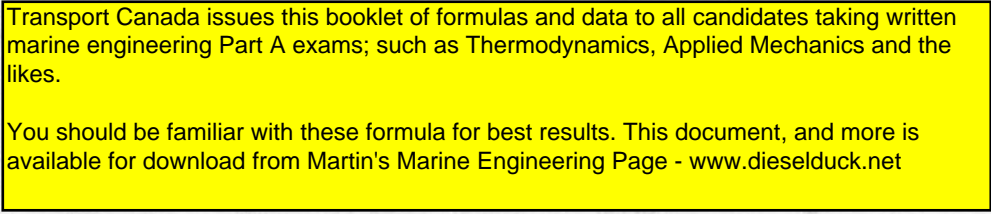
CONTENTS

Engineering Data

Mathematical Formulae	1
Tables	2
Theory of Machines	6
Strength of Materials	9
Thermodynamics	12
Hydrostatics and Hydraulics	15
Electrotechnology	16
Naval Architecture	18
Vapour Tables (R-717 & R - 12)	21

Transport Canada issues this booklet of formulas and data to all candidates taking written marine engineering Part A exams; such as Thermodynamics, Applied Mechanics and the likes.

You should be familiar with these formula for best results. This document, and more is available for download from Martin's Marine Engineering Page - www.dieselduck.net



Engineering Data

Sphere: Surface Area, $A = 4\pi r^2 = \pi d^2$
 Volume, $V = 4/3\pi r^3 = \frac{\pi d^3}{6}$

Cone and Pyramid:

Volume, $V = 1/3$ (area of base) (altitude)

Simpson's Rules:

First Rule: $\Sigma = 1/3 h [y_1 + 4y_2 + 2y_3 + 4y_4 + 2y_5 + \dots + y_n]$
 (odd number of ordinates)

Second Rule: $\Sigma = 3/8 h [y_1 + 3y_2 + 3y_3 + 2y_4 + 3y_5 + 3y_6 + 2y_7 + \dots + y_n]$
 (even number of ordinates)

Factors

$(x+y)^2 = x^2 + 2xy + y^2$
 $(x-y)^2 = x^2 - 2xy + y^2$
 $x^2 - y^2 = (x-y)(x+y)$
 $(x+y)^3 = x^3 + 3x^2y + 3xy^2 + y^3$
 $x^3 + y^3 = (x+y)(x^2 - xy + y^2)$
 $(x-y)^3 = x^3 - 3x^2y + 3xy^2 - y^3$
 $x^3 - y^3 = (x-y)(x^2 + xy + y^2)$

Quadratic Equations

If $ax^2 + bx + c = 0$, then $x = \frac{-b \pm \sqrt{b^2 - 4ac}}{2a}$

Sum of roots = $-\frac{b}{a}$ Product of roots = $\frac{c}{a}$

Also

if $[b^2 - 4ac]$ is
 negative [then roots are [imaginary
 zero [then roots are [real and equal
 positive [then roots are [real and unequal

SI UNIT PREFIXES	Multiples and Submultiples	Prefixes	Symbols
Amount			
1 000 000 000 000	10^{12}	tera	T
1 000 000 000	10^9	giga	G
1 000 000	10^6	mega	M*
1 000	10^3	kilo	k*
100	10^2	hecto	h
10	10	deka	da
0.1	10^{-1}	deci	d
0.01	10^{-2}	centi	c*
0.001	10^{-3}	milli	m*
0.000 001	10^{-6}	micro	μ^*
0.000 000 001	10^{-9}	nano	n
0.000 000 000 001	10^{-12}	pico	p
0.000 000 000 000 001	10^{-15}	femto	f
0.000 000 000 000 000 001	10^{-18}	atto	a

RELATIVE DENSITY OF VARIOUS SUBSTANCES
(ratio between density of substance and density of pure water)

Water (fresh)	1.00	Mica	2.9
Water (sea average)	1.03	Nickel	8.6
Aluminium	2.56	Oil (linseed)	0.94
Antimony	6.70	.. (olive)	0.92
Bismuth	9.80	.. (petroleum)	0.76-0.86
Brass	8.40	.. (turpentine)	0.87
Brick	2.1	Paraffin	0.86
Calcium	1.58	Platinum	21.5
Carbon (diamond)	3.4	Sand (dry)	1.42
Carbon (graphite)	2.3	Silicon	2.3
Carbon (charcoal)	1.8	Silver	10.57
Chromium	6.5	Slate	2.1-2.8
Clay	1.9	Sodium	0.97
Coal	1.36-1.4	Steel (mild)	7.86
Cobalt	8.6	Sulphur	2.07
Copper	8.77	Tin	7.3
Cork	0.24	Tungsten	19.1
Glass (crown)	2.5	Wood (ash)	0.75
Glass (flint)	3.5	.. (beech)	0.7-0.8
Gold	19.3	.. (ebony)	1.1-1.2
Iron (cast)	7.21	.. (elm)	0.66
Iron (wrought)	7.78	.. (lignum-vitae)	1.3
Lead	11.4	.. (oak)	0.7-1.0
Magnesium	1.74	.. (pine)	0.56
Manganese	8.0	.. (teak)	0.8
Mercury	13.6	Zinc	7.0

SPECIFIC HEAT OF LIQUIDS
 $\text{kJ}(\text{kg K})^{-1}$

Alcohol	2.515	Olive Oil	1.96
Ammonia (liquid)	3.98	Petroleum	2.17
Benzol	1.75	Petrol	1.255
Carbonic Acid (liquid)	3.14	Turpentine	1.88
Mercury	0.138	Water	4.187

SPECIFIC HEAT AND LINEAR EXPANSION OF SOLIDS

	Specific Heat $\text{kJ}(\text{kg K})^{-1}$	Mean Expansion per $^{\circ}\text{C}$ $\times 10^{-6}$		Specific Heat $\text{kJ}(\text{kg K})^{-1}$	Mean Expansion per $^{\circ}\text{C}$ $\times 10^{-6}$
Aluminium	0.8876	23	Iron (cast)	0.5443	11
Antimony	0.2009	12	Iron (wrought)	0.4732	12
Bismuth	0.1256	13.5	Lead	0.1298	27.9
Brass	0.3936	18.9	Nickel	0.4563	12.8
Carbon	0.8373	7.92	Platinum	0.1340	8.8
Cobalt	0.4312	12.22	Silicon	0.7076	
Copper	0.3978	16.7	Silver	0.2345	18.7
Glass	0.8290	9	Steel (mild)	0.4857	12
Gold	0.1298	14.4	Tin	0.2345	21.6
Ice	2.110	50	Zinc	0.3894	28.8

MELTING POINT OF VARIOUS SUBSTANCES

Substance	Melting Point °C	Lat. Heat of Fusion kJ/kg	Substance	Melting Point °C	Lat. Heat of Fusion kJ/kg
Aluminium	660	380	Nickel	1453	19
Antimony	630	168	Platinum	1769	114
Bismuth	260	53	Silver	961	104
Brass	80-1000	—	Sulphur	113	39
Chromium	1800	—	Tallow	43	—
Cobalt	1492	—	Tantalum	2996	172
Copper	1083	180	Tin	232	60
Gold	1063	68	Tungsten	3380	—
Glass	1100	—	Zinc	419	100
Ice	0	336	Solder (com.)	171	—
Iron (cast)	1535	109	(2 tin to 1 lead)		
Iron (wrought)	1538	266	Fusible Metal	50	—
Lead	327	23	(5 tin, 3 lead)		
Manganese	1260	—	5 Bismuth	—	—
Mercury	-39	12	3 Mercury)		

PERFECT GASES

At normal atmospheric conditions, and over a limited range of temperature and pressure, the gases listed may be assumed to behave as perfect gases. That is, they may be assumed to have the equation of state $pv = RT$, and to have constant specific heats.

Molar (universal) gas constant: $\bar{R} = MR = 8.3143 \text{ kJ/kmol K}$.

Molar volume of a perfect gas: 1 kmol of any perfect gas occupies a volume of approximately 22.4 m^3 at s.t.p. (0°C and 1 atm).

Gas	Molar mass kg/kmol	Gas constant $\text{kJ}(\text{kg K})^{-1}$	Specific heat capacity $\text{kJ}(\text{kg K})^{-1}$		c_p/c_v
			c_p	c_v	
Air	29.0	0.287	1.01	0.72	1.40
Atmospheric nitrogen †	28.15	0.295	1.03	0.74	1.40
N ₂	28	0.297	1.04	0.74	1.40
O ₂	32	0.260	0.92	0.66	1.40
A	40	0.208	0.52	0.31	1.67
H ₂	2*	4.12	14.20	10.08	1.41
He	4	2.08	5.19	3.11	1.67
CO	28	0.297	1.04	0.74	1.40
CO ₂	44	0.189	0.82	0.63	1.31
SO ₂	64	0.130	0.61	0.48	1.26
CH ₄	16	0.520	2.23	1.71	1.31
C ₂ H ₄	30	0.277	1.75	1.47	1.19
C ₂ H ₆	42	0.198	1.52	1.32	1.15

* A more exact value is 2.016.

† Air contains 0.93% of argon (A) and traces of other gases; these and the nitrogen together are called atmospheric nitrogen.

Real gases are not perfect gases, and the rounded values for R , c_p , c_v and c_p/c_v listed above do not exactly satisfy the relationships between these quantities that would obtain for perfect gases.

Air composition:

Volumetric (and molar): 21.0% O₂, 79.0% atmospheric nitrogen.

Gravimetric: 23.2% O₂, 76.8% atmospheric nitrogen.

<i>A</i>	α	Alpha	<i>N</i>	ν	Nu
<i>B</i>	β	Beta	Ξ	ξ	Xi
Γ	γ	Gamma	<i>O</i>	<i>o</i>	Omicron
Δ	δ	Delta	Π	π	Pi
<i>E</i>	ϵ	Epsilon	<i>P</i>	ρ	Rho
<i>Z</i>	ζ	Zeta	Σ	σ	Sigma
<i>H</i>	η	Eta	<i>T</i>	τ	Tau
Θ	θ	Theta	<i>Y</i>	υ	Upsilon
<i>I</i>	ι	Iota	Φ	ϕ	Phi
<i>K</i>	κ	Kappa	χ	χ	Chi
Λ	λ	Lambda	Ψ	ψ	Psi
<i>M</i>	μ	Mu	Ω	ω	Omega

expansion for common liquids

Liquid	Per °C
Alcohol, methyl	122×10^{-5}
Gasoline	108×10^{-5}
Glycerin	53×10^{-5}
Mercury	18.2×10^{-5}
Petroleum	89.9×10^{-5}
Sulfuric acid	58×10^{-5}
Turpentine	94×10^{-5}
Water (at 20°C)	20.7×10^{-5}

OBSERVABLE	SYMBOL	DIMENSION	MKSC UNIT
Length, position	r, x, ℓ, \dots	L	meter (m)
Time	t, τ	T	second (sec)
Mass	m	M	kilogram (kg)
Area	A, S	L^2	m^2
Volume	V	L^3	m^3
Density	ρ	ML^{-3}	kg/m^3
Frequency	ν, f	T^{-1}	Hertz (Hz) = sec^{-1}
Angle	θ, ϕ	—	radian (rad)
Solid angle	Ω	—	steradian (sr)
Velocity	v, v, w	LT^{-1}	m/sec
Acceleration	a	LT^{-2}	m/sec^2
Angular frequency	ω	T^{-1}	rad/sec
Angular acceleration	α	T^{-2}	rad/sec^2
Force	F	MLT^{-2}	newton (N) = $kg \cdot m/sec^2$
Energy, work	E, W, U, T	ML^2T^{-2}	Joule (J) = N-m
Power	P	ML^2T^{-3}	watt (W) = J/sec
Momentum	p	MLT^{-1}	$kg \cdot m/sec$
Angular momentum	L, J	ML^2T^{-2}	$kg \cdot m^2/sec^2$
Moment of inertia	I	ML^2	$kg \cdot m^2$
Torque	τ	ML^2T^{-2}	N-m
Pressure	P	$ML^{-1}T^{-2}$	$N/m^2 = kg/m \cdot sec^2 = Pa$
Temperature	T	—	degree Kelvin (°K)
Heat	Q	ML^2T^{-2}	J
Entropy	S	$ML^2T^{-2}/°K$	$J/°K$
Electric charge	Q, q	C	coulomb (coul)
Charge density	ρ, ρ_c	CL^{-3}	$coul/m^3$
Surface charge density	σ	CL^{-2}	$coul/m^2$
Current	I	CT^{-1}	ampere (A) = $coul/sec$
Current density	j	$CL^{-2}T^{-1}$	A/m^2
Electrostatic potential	V	$ML^2T^{-2}C^{-1}$	volt (V) = J/coul
Electric field	E	$MLT^{-2}C^{-1}$	$V/m = N/coul$
Polarization	P	CL^{-2}	$coul/m^2$
Resistance	R	$ML^2T^{-3}C^{-2}$	ohm (Ω) = V/A
Resistivity	ρ	$ML^3T^{-3}C^{-2}$	$\Omega \cdot m$
Dielectric constant	K	—	—
Electric dipole moment	p	CL	coul-m
Magnetic field	B	$MT^{-1}C^{-1}$	tesla (T) = $V \cdot sec/m^2$
Magnetic flux	ϕ	$ML^2T^{-1}C^{-1}$	weber (Wb) = $V \cdot sec$
Magnetic dipole moment	m	CL^2T^{-1}	$A \cdot m^2$
Magnetization	M	$CL^{-1}T^{-1}$	A/m
Magnetizing field	H	$CL^{-1}T^{-1}$	A/m
Capacitance	C	$C^2M^{-1}L^{-2}T^2$	farad (F) = $coul/V$
Inductance	L	$ML^2T^{-2}C^{-2}$	henry (H) = $V \cdot sec/A$ = $\Omega \cdot sec$

STRENGTH AND PROPERTIES OF VARIOUS METALS

Metal	Tensile Strength MPa	Compressive Strength MPa	Shear Strength MPa	Modulus of Elasticity GPa	Modulus of Rigidity GPa	Mass Density kg/m ³
Aluminium (cast)	80	—	—	—	—	—
Aluminium (rolled)	124	—	80	—	—	—
Brass (cast)	170	78	93	58	22.5	2574
Brass (wire)	310	78	124	70	26.3	2712
Bronze (phosphor)	—	—	—	62	23.2	8384
Bronze (manganese)	390	—	—	87	34	8384
Copper (cast)	465	620	370	95	35.5	8578
Copper (rolled)	140	—	—	95	35.5	—
Copper (wire)	220	390	170	85	32.5	8772
Copper unannealed (wire)	400	390	186	104	38.5	8772
Copper annealed (wire)	—	—	—	116	46	8772
Gunmetal	280	—	—	108	43.3	8772
Iron (cast)	220	620	186	78	34	8606
Iron (wrought)	125	700	155	124	40.2	7194
Lead	360	340	280	193	77	7775
Muntz Metal	24	46	—	5	1.85	11 400
Steel (mild)	340	360	—	93	34	8190
Steel (tool)	460	465	340	207	82	7830
Steel (castings)	800	800	695	224	85	7830
Tin	390	—	—	185	70	7830
Zinc	30	93	28	39	13.9	7305
	23	230	93	85	32.5	7000

SYMBOLS AND ATOMIC WEIGHTS OF ELEMENTS

Aluminium	Al	27.1	Manganese	Mn	54.9
Antimony	Sb	120.2	Mercury	Hg	200.6
Bismuth	Bi	208.0	Nickel	Ni	58.7
Calcium	Ca	40.0	Nitrogen	N	14.0
Carbon	C	12.0	Oxygen	O	16.0
Chlorine	Cl	35.5	Phosphorus	P	31.0
Chromium	Cr	52.0	Platinum	Pt	195.2
Cobalt	Co	58.9	Potassium	K	39.1
Copper	Cu	63.5	Silicon	Si	28.3
Gold	Au	197.2	Silver	Ag	107.8
Helium	He	4.0	Sodium	Na	23.0
Hydrogen	H	1.0	Sulphur	S	32.0
Iron	Fe	55.8	Tin	Sn	118.7
Lead	Pb	207.2	Tungsten	W	184.0
Magnesium	Mg	24.3	Zinc	Zn	65.3

THEORY OF MACHINES

Uniform Velocity and Acceleration

Linear

$$v_2 = v_1 + at$$

$$v_2^2 = v_1^2 + 2as$$

$$s = \frac{v_1 + v_2}{2} t$$

$$s = v_1 t + \frac{1}{2} at^2$$

Angular

$$\omega_2 = \omega_1 + \alpha t$$

$$\omega_2^2 = \omega_1^2 + 2\alpha\theta$$

$$\theta = \frac{\omega_1 + \omega_2}{2} t$$

$$\theta = \omega_1 t + \frac{1}{2} \alpha t^2$$

s = Linear Distance m θ = Angular Displacement radians
 v = Linear Velocity m/s ω = Angular Velocity rad/s
 a = Linear Acceleration m/s² α = Angular Acceleration rad/s²
 t = Time s t = Time s
 Suffices 1 and 2 refer to initial and final conditions respectively.

Relationship between linear and angular quantities

$$s = \theta r$$

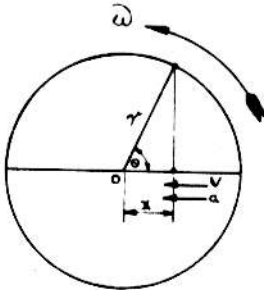
$$v = \omega r$$

$$a = \alpha r$$

Motion of a point travelling on a circular path

$$\text{Centripetal acceleration} = \omega^2 r = \frac{v^2}{r}$$

Circle Diagram



$$v = \omega r \sin \theta = \omega \sqrt{r^2 - x^2}$$

$$a = \omega^2 r \cos \theta = \omega^2 x$$

x = Displacement from mean position.
 r = Amplitude or motion.
 ω = Uniform Angular Velocity of the point travelling on the circumference of the circle.

$$\text{Periodic time} = \frac{2\pi}{\omega}$$

$$\text{Frequency} = \frac{1}{\text{periodic time}}$$

$$\text{Periodic time} = 2\pi \sqrt{\frac{\text{displacement}}{\text{acceleration}}}$$

Dynamics

Fundamental force equation

$$F = ma$$

Fundamental torque equation

$$T = J\alpha \text{ where } J = mk^2$$

Linear momentum = mv

Angular momentum = $J\omega$

Impulse = Force \times Time

$$= \text{Mass} \times \text{Change of velocity}$$

$$= \text{Change of momentum}$$

Impulse torque = Torque \times Time

$$= \text{Mass moment of inertia} \times \text{Change of angular velocity}$$

$$= \text{Change of angular momentum}$$

Coefficient of restitution

$$\text{between two colliding bodies} = \frac{\text{Relative velocity between bodies after impact}}{\text{Relative velocity between bodies before impact}}$$

Kinetic energy of translation = $\frac{1}{2}mv^2$

Kinetic energy of rotation = $\frac{1}{2}J\omega^2 = \frac{1}{2}mk^2\omega^2$

Work done by a force = $\int F.ds$

$$= Fs \text{ if force is constant}$$

Work done by a torque = $\int Td\theta$

$$= T\theta \text{ if torque is constant}$$

Power developed by a force = Fdv

$$= Fv \text{ if force and velocity are constant}$$

Power developed by a torque = $T.d\omega$

$$= T\omega \text{ if torque and angular velocity are constant}$$

Centrifugal force

$$= m\omega^2 r = \frac{mv^2}{r}$$

F = unbalanced acceleration force

(newtons N)

T = unbalanced accelerating torque

(Nm)

m = mass

(kg)

a = linear acceleration

(m/s^2)

α = angular acceleration

(rad/s^2)

J = mass moment inertia about axis of rotation

($kg.m^2$)

$$= mk^2$$

k = radius of gyration about axis of rotation

(m)

s = linear displacement

(m)

θ = angular displacement

(radians)

v = linear velocity

(m/s)

ω = angular velocity

(rad/s)

Energy units in joules(J)

Power units in watts (W)

Gyroscopic Torque

$$T = J\omega\Omega = mk^2\omega\Omega$$

J = mass moment of inertia of wheel

ω = angular velocity of wheel

Ω = angular velocity of precession

Friction

$$F = \mu N \text{ where } \mu = \frac{F}{N} = \tan \phi$$

F = friction force

N = reaction normal to surface

μ = coefficient of friction

ϕ = friction angle

Inclined Plane

$$\begin{aligned} \text{Force parallel to plane} &= W \sin \theta \text{ (without friction)} \\ &= \frac{W \sin (\theta + \phi)}{\cos \phi} \text{ (with friction)} \\ \text{Force parallel to base} &= W \tan \theta \text{ (without friction)} \\ &= W \tan (\theta + \phi) \text{ (with friction)} \\ \text{Force at angle to } \alpha \text{ plane} &= \frac{W \sin \theta}{\cos \alpha} \text{ (without friction)} \\ &= \frac{W \sin (\theta + \phi)}{\cos (\alpha - \phi)} \text{ (with friction)} \end{aligned}$$

where W = weight of body on plane
 θ = angle of plane
 ϕ = friction angle
 Force = force to move body up plane.

Threads (With Friction)

$$\begin{aligned} \text{Square} \quad \eta &= \frac{\tan \theta}{\tan (\theta + \phi)} & \theta &= \text{thread angle} \\ & & \phi &= \text{friction angle} \\ \eta_{\max} &= \frac{1 - \sin \phi}{1 + \sin \phi} \end{aligned}$$

Single Plate Clutch

$$\text{Torque transmitted} = \frac{2}{3} \mu W \left(\frac{R_1^3 - R_2^3}{R_1^2 - R_2^2} \right)$$

Uniform pressure theory

Torque transmitted = $\mu W R_m$
 For multi-plate clutch with n pairs of contact surfaces torque transmitted will be n times the above.

Uniform wear theory

Cone Clutch

$$\text{Torque transmitted} = \frac{\mu W}{\sin \theta} \left(\frac{R_1^3 - R_2^3}{R_1^2 - R_2^2} \right)$$

Uniform pressure theory

$$\text{Torque transmitted} = \mu W R_m$$

Uniform wear theory

Unless otherwise stated use uniform pressure theory for bearings, uniform wear theory for clutches.

R_m = mean radius
 W = Axial load
 θ = Semi apex angle.

Centrifugal Clutch

$$\text{Torque transmitted} = n \mu R (F - P)$$

n = Number of shoes

R = Inside radius of rim

F = Centrifugal force on each shoe

P = Force exerted on each shoe by retaining spring.

Belt Drives

$$\text{Flat Belts.} \quad \frac{T_1 - T_c}{T_2 - T_c} = e^{\mu \theta}$$

T_1 = Tension on tight side of belt

(N) $T_1 - T_c$ = Effective tension tight side.

T_2 = Tension on slack side of belt

(N) $T_2 - T_c$ = Effective tension slack side.

T_c = Centrifugal tension

$$T_c = m v^2$$

m = mass per unit length of belt (kg/m)

v = linear velocity of belt (m/s)

$$\text{Power transmitted} = (T_1 - T_2)v$$

Maximum power transmitted when $T_c = \frac{T_1}{3}$

$$\text{Initial tension } T_o = \frac{T_1 + T_2}{2}$$

STRENGTH OF MATERIALS

<p>A = Area $A\bar{x}$ = 1st moment of area D, d = diameter E = Young's Modulus of Elasticity F = Shearing force G = Modulus of Rigidity or Shear h = Height, distance I = 2nd moment of area J = Polar 2nd moment of area K = Bulk Modulus L = Length M = Bending moment P = Force p = Pressure, compressive stress R, r = Radius T = Torque U = Strain energy W = Load, weight</p>	<p>Δ = Deflection ϵ = Direct strain θ = Twist, angle γ = Shear strain ρ = Density σ = Direct stress τ = Shear stress ν = Poisson's Ratio ω = Angular velocity</p>
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Stress, Strain and Moduli

$$\text{Modulus of Elasticity } (E) = \frac{\text{direct stress } (\sigma)}{\text{direct strain } (\epsilon)} = \frac{\frac{\text{load}}{\text{area}}}{\frac{\text{extension}}{\text{original length}}}$$

$$\text{Poisson's Ratio } \nu = \frac{\text{lateral strain}}{\text{linear strain}}$$

$$\text{General expression for strain } \epsilon_x = \frac{\sigma_x}{E} - \nu \frac{\sigma_y}{E} - \nu \frac{\sigma_z}{E}$$

$$\text{Modulus of Rigidity } (G) = \frac{\text{shear stress } (\tau)}{\text{shear strain } (\gamma)}$$

$$\text{Bulk Modulus } (K) = \frac{\text{fluid press } (p)}{\text{volumetric strain } (\epsilon_v)}$$

$$G = \frac{E}{2(1 + \nu)}$$

$$K = \frac{E}{3(1 - 2\nu)}$$

$$\text{Principal Stress} = \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau^2}$$

$$\text{Angle of Principal planes: } \tan 2\theta = \frac{2\tau}{\sigma_x - \sigma_y}$$

$$\text{Maximum Shear Stress} = \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau^2}$$

$$\text{Principal Strains} = \frac{\epsilon_x + \epsilon_y}{2} \pm \sqrt{\left(\frac{\epsilon_x - \epsilon_y}{2}\right)^2 + \frac{\gamma^2}{4}}$$

Strain Energy

$$\text{Direct stress } U = \frac{\sigma^2}{2E} \times \text{volume}$$

$$\text{Shear stress } U = \frac{\tau^2}{2G} \times \text{volume}$$

Beams and Bending

$$\frac{M}{I} = \frac{\sigma}{y} = \frac{E}{R}$$

- M = Bending moment
- I = Second moment of area
- σ = Stress
- y = Distance from neutral axis.
- E = Modulus of elasticity
- R = Radius of curvature.

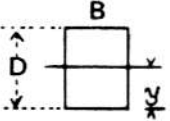

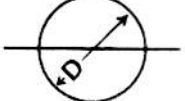
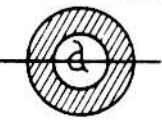
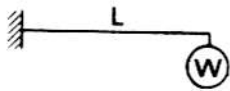
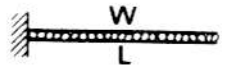
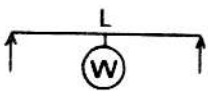
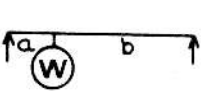
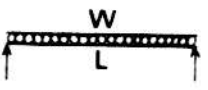
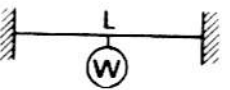
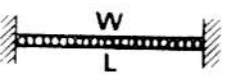
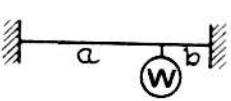
Section	I	r	$\frac{I}{r^2}$
	$\frac{BD^3}{12}$	$\frac{D}{2}$	$\frac{BD^2}{6}$
	$\frac{BD^3 - bd^3}{12}$	$\frac{D}{2}$	$\frac{BD^3 - bd^3}{6D}$
	$\frac{\pi}{64} D^4$	$\frac{D}{2}$	$\frac{\pi}{32} D^3$
	$\frac{\pi}{64} (D^4 - d^4)$	$\frac{D}{2}$	$\frac{\pi}{32} \left(\frac{D^4 - d^4}{D} \right)$

TABLE OF BENDING MOMENTS AND DEFLECTIONS

Load Diagram	Bending Moment	Deflection
	WL (at wall)	$\frac{WL^3}{3EI}$ (at free end)
	$\frac{WL}{2}$ (at wall)	$\frac{WL^3}{8EI}$ (at free end)
	$\frac{WL}{4}$ (at centre)	$\frac{WL^3}{48EI}$ (at centre)
	$\frac{Wab}{a+b}$ (at load)	$\frac{Wa}{3LEI} \left\{ \frac{b(2a+b)}{3} \right\}^{\frac{3}{2}}$ at $\left\{ \frac{b(2a+b)}{3} \right\}^{\frac{1}{2}}$ from right
	$\frac{WL}{8}$ (at centre)	$\frac{5WL^3}{384EI}$ (at centre)
	$\frac{WL}{8}$ at centre and ends	$\frac{WL^3}{192EI}$ (at centre)
	$\frac{WL}{12}$ (at the ends) $\frac{WL}{24}$ (at the centre)	$\frac{WL^3}{384EI}$ (at centre)
	$\frac{Wab^2}{L^2}$ (at 'a' end) $\frac{Wba^2}{L^2}$ (at 'b' end)	$\frac{\frac{2}{3}Wa^3b^2}{(3a+b)^2EI}$ (maximum where a is greater than b)

Torsion

general formula $\frac{T}{J} = \frac{\tau}{r} = \frac{G\theta}{l}$

power transmitted = $T\omega = \frac{2\pi NT}{60}$ watts

twist on solid tapered shaft = $\frac{32Tl}{3\pi G} \left(\frac{D_1^2 + D_1 D_2 + D_2^2}{D_1 D_2} \right)$

$N = \text{rev/min}$
 $T = [\text{N m}]$

where

- $D_1 = \text{small diameter}$
- $D_2 = \text{large diameter}$
- $l = \text{length}$

Equivalent Bending Moment or Equivalent Torque on a Shaft

$M_E = \frac{M + \sqrt{M^2 + T^2}}{2}$

$T_E = \sqrt{M^2 + T^2}$

Principal Stress

Maximum Shear Stress

Moments of Inertia and k^2

Sketch	Description	Second Moments	(Radius of Gyration) ²
	Rectangle about axis yy'	$\frac{BD^3}{12}$	$\frac{D^2}{12}$
	Rectangle about axis zz'	$\frac{DB^3}{12}$	$\frac{B^2}{12}$
	Rectangle about axis OO	Area $\times k^2 =$ $BD \left[\frac{D^2 + B^2}{12} \right]$	$\frac{D^2 + B^2}{12}$
	Rectangle about external axis x from C of G	$BD \left[\frac{D^2}{12} + x^2 \right]$	$\left(\frac{D^2}{12} + x^2 \right)$
	For rectangular solids (k^2) remains the same but Vol is substituted for Area in expression for 1:		
	Rectangular solid rotating about external axis OO which is x from C of G	Vol $\frac{D^2 + B^2}{12} + x^2$	$\frac{D^2 + B^2}{12} + x^2$
	Circle about a diameter	$\frac{\pi D^4}{64}$	$\frac{D^2}{16} = \frac{R^2}{4}$
	Circle about its centre	$\frac{\pi D^4}{32}$	$\frac{D^2}{8} = \frac{R^2}{2}$
	Cylinder about its axis	Vol $\left[\frac{D^2}{8} \right]$	$\frac{D^2}{8} = \frac{R^2}{2}$
	Sphere rotating about a diameter	Vol $\left[\frac{D^2}{10} \right]$	$\frac{D^2}{10} = \frac{3}{5} R^2$
	Cylinder about an axis at right angles to axis but through the C of G	Vol $\left[\frac{D^2}{16} + \frac{L^2}{12} \right]$	$\frac{D^2}{16} + \frac{L^2}{12}$

ENGINEERING THERMODYNAMICS

Symbols Used

T	= Thermodynamic temperature	(K)
θ	= Temperature value	(C)
T_f	= Saturation temperature	
P, p	= Pressure	
V, v	= Volume	
m	= Mass	
ρ	= Density (mass/unit volume)	
W	= Work energy	
P	= Power	
η	= Efficiency	
n	= Amount of substance (moles or kilomoles)	
V_o	= Molar volume	
C	= Vapour velocity	
Q	= Quantity of heat, Q_s = Sensible heat	
H	= Enthalpy, h = specific enthalpy (enthalpy per unit mass)	
U	= Internal energy (molecular), u = specific internal energy	
E	= Total internal energy including kinetic energy (K.E.) and/or P.E.	
e	= Specific total internal energy	
h_f	= Specific liquid enthalpy	
h_{fg}	= Specific enthalpy of evaporation	
h_g	= Specific enthalpy of dry saturated vapour	
c	= Specific heat capacity	
cp	= Specific heat at constant pressure	
cv	= Specific heat at constant volume	
R	= Specific gas constant	
S	= Entropy, s = specific entropy	
x	= Dryness fraction	
R_o	= Universal gas constant	
ϕ	= Heat flow rate	
λ	= Thermal conductivity	
α	= Coefficient, linear expansion	
β	= Coefficient, cubical expansion.	

RELATIONSHIPS BETWEEN TEMPERATURE AND VOLUME, AND TEMPERATURE AND PRESSURE, when $pV^n = C$

$$\frac{T_1}{T_2} = \left\{ \frac{V_2}{V_1} \right\}^{n-1} = \left\{ \frac{p_1}{p_2} \right\}^{\frac{n-1}{n}}$$

The perfect gas

The characteristic equation of state

$$pv = RT$$

or for m kg, occupying V m³,

$$pV = mRT$$

Universal Gas Equation

$$PV = nR_o T$$

$R_o = MR = 8.3143$ kJ/kg mol. K

M = mass of one mol. in kg

n = number of mols

Work Formulae

(i) $PV = \text{Constant}$

$$W = PV \log_e \frac{V_2}{V_1}$$

(ii) $PV^n = \text{Constant}$

$$W = \frac{(P_1 V_1 - P_2 V_2)}{n-1}$$

(iii) $P = \text{Constant}$

$$W = P(V_2 - V_1)$$

General

The efficiency of any heat engine cycle is given by

$$\eta = \frac{Q_1 - Q_2}{Q_1} \quad \text{where } Q_1 \text{ is the sum of the heat flows to the cycle and } Q_2 \text{ the sum of the heat flows from the cycle.}$$

Carnot

$$\eta = \frac{T_1 - T_2}{T_1} \quad \text{for any reversible cycle working between one heat source and one heat sink.}$$

Constant Volume Air Standard Cycle (A.S.E.)

$$\eta = 1 - \frac{1}{r_v^{\gamma-1}} \quad r_v = \text{volume compression ratio.}$$

Constant Pressure Cycle (Joule)

$$\eta = 1 - \frac{T_4 - T_3}{T_3 - T_2} = 1 - \frac{1}{r_v^{\gamma-1}}$$

$r_v = \text{volume compression ratio}$
 $T_1 = \text{temperature at beginning of compression.}$

Diesel Cycle

$$\eta = 1 - \frac{1}{\gamma} \left[\frac{T_4 - T_1}{T_3 - T_2} \right]$$

$$= 1 - \frac{1}{r_v^{\gamma-1}} \left[\frac{r_c^\gamma - 1}{\gamma(r_c - 1)} \right]$$

$r_v = \text{volume compression ratio}$
 $r_c = \frac{T_3}{T_2}$
 $T_1 = \text{temperature at beginning of compression.}$

Dual Cycle

$$\eta = 1 - \frac{1}{r_v^{\gamma-1}} \cdot \frac{r_p r_c^\gamma - 1}{(r_p - 1) + \gamma r_p (r_c - 1)}$$

$r_v = \text{vol. compression}$
 $r_c = \frac{V_4}{V_3} \text{ ratio}$
 $r_p = \frac{p_3}{p_2}$
 $v_1 = \text{beginning of compression}$

Heat Transfer

(i) Conduction

for steady heat flow through a flat surface of one material,

$$Q = \frac{kA(t_1 - t_2)}{x}$$

steady flow through a flat composite wall,

$$Q = \frac{A(t_1 - t_n)}{\sum \frac{x}{k}}$$

(always carefully note the units in which a 'k' value is given)

through a thick cylinder,

$$Q = \frac{2\pi l k (t_1 - t_2)}{\log_e \frac{r_2}{r_1}}$$

through a composite thick cylinder,

$$Q = \frac{l(t_1 - t_n)}{\sum \left[\frac{\log_e \frac{r^{n+1}}{r^n}}{2\pi k_n} \right]}$$

t_1 are surface temperatures of the wall or plate.

A = surface area m^2

x = thickness (m)

k = conductivity coefficient
(Watts/ $m^\circ C_0$)

l = length (m)

r = inner radius

r_2 = outer radius (m)

t = inner temperature

t_n = outer temperature

r^{n+1} and r^n are the radii of any included cylinder and k_n its conductivity coefficient.

(ii) Convection

The overall heat transfer coefficient is U .

$$\frac{1}{U} = \frac{1}{h_1} + \frac{1}{k/x} + \frac{1}{h_2}$$

$$Q = UA(t_1 - t_4) = h_1 A(t_1 - t_2) = h_2 A(t_3 - t_4) = kA \frac{(t_2 - t_3)}{x}$$

t_1 = temperature of hot fluid

t_2 = temperature of tube wall on hot fluid side

t_3 = temperature of tube wall on cool side

t_4 = temperature of cool fluid

Heat Exchangers

In these the overall temperature difference across the wall varies and a mean temperature difference is used in the Q equation.

mean temperature difference for parallel flow and contra flow exchangers

$$= \frac{\Delta t_1 - \Delta t_0}{\log_e \left(\frac{\Delta t_1}{\Delta t_0} \right)}$$

Δt_1 = temp. difference at inlet end of hot fluid.

Δt_0 = temp. difference at outlet end of hot fluid.

Radiation

Stefan-Boltzmann Law

for 'black' body $q_b = \sigma T^4$

q_b = energy emitted per unit area per unit time, for black body

σ = Stefan-Boltzmann constant

T = absolute temperature (K)

for 'grey' body $\epsilon = \frac{q}{q_b}$

$$q = \epsilon q_b = \epsilon \sigma T^4$$

q = energy emitted (or radiated) per unit area per unit time for 'grey' body

ϵ = emissivity, being unity for black body and less than unity for any real body

$$\epsilon = \alpha$$

for 'grey' body of surface area A and at absolute temperature T_1 , and surroundings at absolute temperature T_2 , we have

α = ratio of energy absorbed by a real body to energy absorbed by a 'black' body

$$A = m^2$$

$$q \text{ radiated} = \epsilon A \sigma T_1^4$$

$$q \text{ absorbed} = \alpha A \sigma T_2^4$$

$$\text{nett } q = \epsilon A \sigma T_1^4 - \alpha A \sigma T_2^4$$

$$= \epsilon A \sigma (T_1^4 - T_2^4)$$

$$= 56.7 \times 10^{-12} \epsilon A (T_1^4 - T_2^4) \text{ kJ/s}$$

Substance	Thermal conductivity $W(m K)^{-1}$
Pure copper	386
Pure aluminium	229
Duralumin	164
Cast iron	52
Mild steel	48.5
Lead	34.6
Concrete	0.85 to 1.4
Building brick	0.35 to 0.7
Wood (oak)	0.15 to 0.2
Rubber	0.15
Cork board	0.043

Typical Values of ϵ	
Copper, polished	0.02
Copper, dull	0.56
Steel, polished	0.07
Steel, dull	0.8
White paint	0.96
Lamp black	0.96

HYDROSTATICS AND HYDRAULICS

Pressure at Depth h from Free Surface

$$\text{Pressure} = h\rho g \text{ N/m}^2$$

where h = depth to point considered in metres

$$\rho = \text{fluid density in kg/m}^3$$

$$g = 9.81 \text{ m/s}^2$$

Total Pressure Load = $\bar{h}a\rho g$ Newtons on immersed area

where a is area considered in m^2

and \bar{h} is depth from free surface to centroid of figure in metres.

Centre of Pressure

$$\text{Depth to C of P from free surface} = \frac{\text{2nd moment of area about surface}}{\text{1st moment of area about surface}}$$

Common Cases

Rectangle, depth D , top edge in surface: C of P at $\frac{7}{8}D$ depth

Circle, diameter D , top edge in surface: C of P at $\frac{5}{8}D$ depth

Triangle, vertical depth H , side in surface: C of P at $\frac{1}{2}H$ depth

Triangle, vertical depth H , corner in surface: C of P at $\frac{3}{4}H$ depth.

Energy Equation

Expressed per unit *Weight* of fluid flowing.

$$H = Z + \frac{P}{\rho g} + \frac{v^2}{2g} + \text{losses}$$

where H = total head in metres

Z = height above datum level in metres

P = pressure in N/m^2

ρ = fluid density in kg/m^3

v = fluid velocity in m/s

$$g = 9.81 \text{ m/s}^2$$

loss may be due to any or all of the following:

$$\text{loss due to enlargement} = \frac{(v_1 - v_2)^2}{2g}$$

$$\text{loss due to contraction} = \frac{k v^2}{2g} \text{ usually taken as } \frac{0.5 v^2}{2g}$$

$$\text{loss due to friction hf} = \frac{4fLv^2}{2gd}$$

where f is a constant, depending on the type of flow and the pipe finish, d is the pipe diameter and L the length of pipe considered in metres.

Orifices

Theoretical velocity of flow from an orifice = $\sqrt{2gh}$ m/s

where h is depth to centre of orifice

Actual velocity of flow = $C_v \sqrt{2gh}$ m/s

small round orifices $C_v \approx 0.97$

Area of jet at Vena Contracta = $C_a \times \text{Area Orifice}$

average value of C_a for small round orifices = 0.64

Quantity flowing from an orifice = $C_d \times \text{Area Orifice} \times \sqrt{2gh}$

where,

area orifice is in m^2

h is in m

C_d is the coefficient of discharge = $C_v \times C_a$

Velocity of a Pressure Wave or Sound in a Liquid

$$\alpha = \frac{k}{\rho}$$

where k is the coefficient of compressibility of the liquid in N/m^2

ρ is the liquid density kg/m^3

$\alpha \approx 1430 \text{ m/s}$ for water.

ELECTROTECHNOLOGY

<i>Electrical Quantity</i>	<i>Symbol</i>	<i>Unit</i>	<i>Unit Abbrevia- tion</i>
Admittance	Y	siemens	S
Angular velocity	ω	radian per second	rad/s
Capacitance	C	farad microfarad	F μ F
Charge or Quantity of electricity	Q	coulomb	C
Conductance	G	siemens	S
Conductivity	σ	siemens per metre	S/m
Current			
Steady or r.m.s. value	I	ampere milliampere microampere	A mA μ A
Instantaneous value	i		
Maximum value	I_m		
Current density	J	ampere per metre ²	A/m ²
Difference of potential			
Steady or r.m.s. value	V	volt millivolt kilovolt	V mV kV
Instantaneous value	v		
Maximum value	V_m		
Electric force (Electric field strength)	E	volt per metre	V/m
Electric flux	ψ	coulomb	C
Electric flux density	D	coulomb per square metre	C m ⁻²
Electromotive force			
Steady or r.m.s. value	E	volt	V
Instantaneous value	e		
Maximum value	E_m		
Energy	W	joule watt-hour kilowatt-hour electronvolt	J Wh kWh eV
Frequency	f	hertz kilohertz megahertz	Hz kHz MHz
Impedance	Z	ohm	Ω
Inductance, self	L	henry (plural, henrys)	H
Inductance, mutual	M	henry (plural, henrys)	H
Magnetizing force (Magnetic field strength)	H	ampere per metre	A/m
Magnetic flux	ϕ	weber	Wb
Magnetic flux density	B	tesla	T
Magnetomotive force	F	ampere	A
Permeability of free space	μ_0	henry per metre	H/m
Permeability, relative	μ_r		
Permeability, absolute	μ	henry per metre	H/m
Permittivity of free space	ϵ_0	farad per metre	F/m
Permittivity, relative (Di-electric constant)	ϵ_r		
Permittivity, absolute	ϵ	farad per metre	F/m
Power	P	watt kilowatt megawatt	W kW MW
Reactance	X	ohm	Ω
Reactive voltampere		var	VA _r
Reluctance	S	ampere per weber	A/Wb
Resistance	R	ohm microhm megohm	Ω $\mu\Omega$ M Ω
Resistivity	ρ	ohm-metre microhm-metre ohm-centimetre	Ω m $\mu\Omega$ m Ω cm
Susceptance	B	siemens	S
Voltampere		voltampere kilovoltampere	VA kVA

ELECTRICAL CIRCUITS

Simple D.C. Circuit (steady conditions)

(a) Steady Conditions

Ohm's Law $I = \frac{V}{R}$

Power in circuit $= IV = I^2 R = \frac{V^2}{R}$ watts

Energy in time $t = Vt$ joules

For resistances in series $R = R_1 + R_2 + R_3 + \dots$

For resistances in parallel $\frac{1}{R} = \frac{1}{R_1} + \frac{1}{R_2} + \frac{1}{R_3} + \dots$

that is $G = G_1 + G_2 + G_3 + \dots$

Resistivity $R = \frac{\rho l}{a}$

Temperature effects $R = R_0(1 + \alpha t)$

Induced e.m.f. $= -L \frac{di}{dt}$ volts.

(b) Inductive Conditions (e.g. switching)

i (rising) $= I(1 - e^{-\frac{Rt}{L}}) = I(1 - e^{-\frac{t}{T}})$

i (decaying) $= Ie^{-\frac{Rt}{L}}$

Time constant $T = \frac{L}{R}$ seconds

Energy stored in magnetic field $= \frac{1}{2}LI^2$ joules.

Simple a.c. Circuit

$e = E \sin \theta = 2\pi BAN n \sin \theta$ volts

Average current $= \frac{\sum i}{\text{number of ordinates}} = 0.637 I_m$ if sinusoidal

r.m.s. current $= \sqrt{\frac{\sum i^2}{\text{number of ordinates}}} = 0.707 I_m$, if sinusoidal

form factor $= \frac{\text{r.m.s. value}}{\text{average value}} = 1.11$, if sinusoidal.

Circuit Laws

(a) Pure Resistance Current

$I = \frac{V}{R}$ current and voltage are in phase

(b) Pure Inductive Current

$I = \frac{V}{2\pi fL} = \frac{V}{X_L}$ current lags applied voltage by 90° ($\phi = 90^\circ$)

(c) Pure Capacitive Current

$I = 2\pi fCV$ current leads applied voltage by 90° ($\phi = 90^\circ$)

$= \frac{V}{\frac{1}{2\pi fC}} = \frac{V}{X_C}$

(d) RLC Series Circuit

$I = \frac{V}{Z}$ where $Z = \sqrt{R^2 + \left(2\pi fL - \frac{1}{2\pi fC}\right)^2}$

$= \sqrt{R^2 + (X_L - X_C)^2}$

for resonance, $X = X_C$, $f = \frac{1}{2\pi \sqrt{LC}}$

(e) RLC Parallel Circuit

$C = \frac{L}{R^2 + (2\pi fL)^2}$

$f = \frac{1}{2\pi \sqrt{LC}}$ provided $R \ll 2\pi fL$

and, impedance at resonance $= \frac{L}{CR}$

NAVAL ARCHITECTURE

Notation

Z	= chosen ordinate in integration rules
d	= draught, or distance
∇	= displacement volume
Δ	= ship mass
W	= ship weight
w	= weight (general)
v	= volume of emerged/immersed wedge
S	= wetted surface
A_w	= waterplane area
B	= centre of buoyancy (also beam)
L	= ship length (general)
M	= metacentre
G	= centre of ship mass
BM	= metacentric radius
GM	= metacentric height
GZ	= righting lever
G_1G_2	= movement of G
I_t	= second moment of area of waterplane about a transverse axis through centroid
i	= second moment of area of a free liquid surface about a longitudinal axis through its centroid
ρ	= density
MCT 1 cm	= moment to change trim one cm
R_f	= frictional resistance
R_r	= residual resistance
R_t	= total resistance
V	= hull speed
g	= acceleration due to gravity (also centroids of immersed/emerged wedges)
f	= Froude friction factors
R_n	= $\frac{VL}{v}$ (Reynolds number) non-dimensional
F_n	= $\frac{V}{\sqrt{Lg}}$ (Froude's number) non-dimensional
T	= thrust
t	= thrust deduction factor
a	= augment of resistance factor
w_t	= Froude wake fraction
w_i	= Taylor wake fraction
P_D	= delivered power
P_E	= effective power
P_T	= thrust power
η_o	= open water propeller efficiency
η_B	= behind hull propeller efficiency
η_r	= $\frac{\eta_B}{\eta_o}$ = relative rotative efficiency
η_H	= hull efficiency
QPC	= quasi propulsive coefficient
P	= propeller pitch
N	= revolutions per minute
n	= revolutions per second
V_a	= speed of advance of propeller
s	= propeller slip ratio
D	= propeller diameter
Q	= propeller torque

TONNE PER CENTIMETRE IMMERSION :

$$TPC = \frac{A_w \times \rho}{100}$$

Transverse Stability

Morrish Formula

for approximate position of centre of buoyancy B

$$\text{from waterline} = \frac{1}{3} \left(\frac{d}{2} + \frac{\nabla}{A} \right)$$

$$\text{from keel, } KB = \frac{1}{3} \left(\frac{5d}{2} - \frac{\nabla}{A} \right)$$

Metacentric Radius

$$BM = \frac{I_f}{\nabla}$$

Moment of Statical Stability

$$= W \times GZ \text{ or } \Delta \rho GZ$$

$$\text{for small angles } GZ = GM \sin \theta = GM \theta_c$$

$$\text{for large angles } GZ = \frac{v \times hh}{\nabla} - BG \sin \theta$$

where

$v \times hh$ = horizontal transfer moment of wedges.

Free Surface Effect

$$G_1 G_2 = \frac{i}{\nabla} \times \frac{\rho_F}{\rho_s}$$

ρ_F = density of free liquid
 ρ_s = sea density

Inclining Experiment

$$GM = \frac{w \times d}{W \tan \theta} \text{ or } \frac{m \times d}{\Delta \tan \theta}$$

w = weight moved
 d = distance moved
 W = ship weight
 θ = angle of heel

Statical Stability Curve

A curve of GZ against θ as ship heels at constant displacement.
correction for movement of $G = G_1 G_2 \sin \theta$ along ship centreline
correction for movement of $G = G_1 G_2 \cos \theta$ parallel to deck.

Dynamical Stability

$$= W \int^{\theta} GZ d\theta = W \times \text{area under } GZ \text{ curve up to that angle of heel}$$

$$= W^0 \left[\frac{v(gh+gh)}{\nabla} - BG(1 - \cos \theta) \right] \text{ where } v(gh+gh) \text{ is the vertical transfer moment (volume) of the wedges}$$

Longitudinal Stability

$$BM_L = \frac{I_f}{\nabla}$$

where d is distance of centre of flotation (centroid) from midships

$$I_f = I_x - Ad^2$$

$$\text{MCT 1 cm} = \frac{\Delta GM_L}{100L} \text{ tonne metre}$$

Δ = tonnes
 GM_L = metres
 L = metres

Note. BM_L may be used in this formula without serious error.

Resistance
 $R_f = fSV^{1.825}$ newtons

f = Froude friction factor
 $S = m^2$
 $V =$ knots (1852 m/h)

$R_T = R_F + R_R$

Corresponding Speeds Law

$\frac{V}{\sqrt{L}}$ has same value for both hulls.

Law of Comparison

$\frac{R_R}{W}$ has same value for both hulls at corresponding speeds and draughts

$\therefore \frac{R_{R1}}{R_{R2}} = \frac{W_1}{W_2} = \frac{\Delta_1 g}{\Delta_2 g} = \frac{\rho_1 \nabla_1}{\rho_2 \nabla_2} = \frac{\rho_1}{\rho_2} \times \text{scale}^3$

Modern Coefficients

$C_T = C_F + C_R$

at corresponding speeds $C_{R(\text{ship})} = C_{R(\text{model})}$

$C_T = \frac{R_T}{\frac{1}{2}\rho SV^2}$
 $C_F = \frac{R_F}{\frac{1}{2}\rho SV^2}$
 $C_R = \frac{R_R}{\frac{1}{2}\rho SV^2}$

Modern Friction Lines

These are empirical formulae derived from plank experiments. They express C_F in terms of Reynold's number (R_n). One such formulae (ITTC) is:

$C_F = \frac{0.075}{(\log R_n - 2)^2}$

Note a correction factor ΔC_f is added when using these plank formulae to calculate C_F for a ship—to allow for scale, roughness and form effects.

Propulsion

(true) $s = \frac{PN - V_a}{PN}$
 (apparent) $s = \frac{PN - V_s}{PN}$ } use consistent units.

$P_{E(KW)} = R_{(KN)} V_{(m/s)}$

$P_{T(KW)} = T_{(KN)} V_a(m/s)$

$P_{D(KW)} = 2\pi N(\text{rev/s}) Q(\text{KN m})$

$w_r = \frac{V_s - V_a}{V_a}$ $w_t = \frac{V_s - V_a}{V_s} = \frac{w_r}{1 + w_r}$

$T - R_T = iT = aR_T$

hull efficiency = $\frac{P_E}{P_T} = \frac{R_T V_s}{T V_a} = \frac{1 + w_r}{1 + a} = (1 + w_r)(1 - t)$

propeller efficiency = $\frac{P_T}{P_D}$

relative rotative efficiency = $\frac{\eta_B}{\eta_0}$ (= R.R.E.)

QPC = $\frac{P_E}{P_D}$ = hull efficiency \times open water efficiency \times RRE

PC = QPC \times transmission efficiency = $\frac{P_E}{\text{Engine power}}$

